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SAMARA UNIVERSITY

Determination of Relative Position and Orientation of Nanosatellites by Video Image Analysis

Postgraduate of Space Research Department
Sergey Simakov

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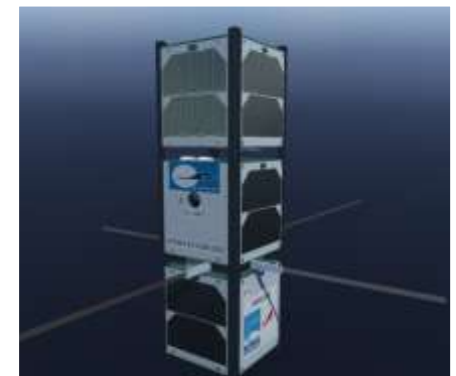




Teaching students through design nanosatellites

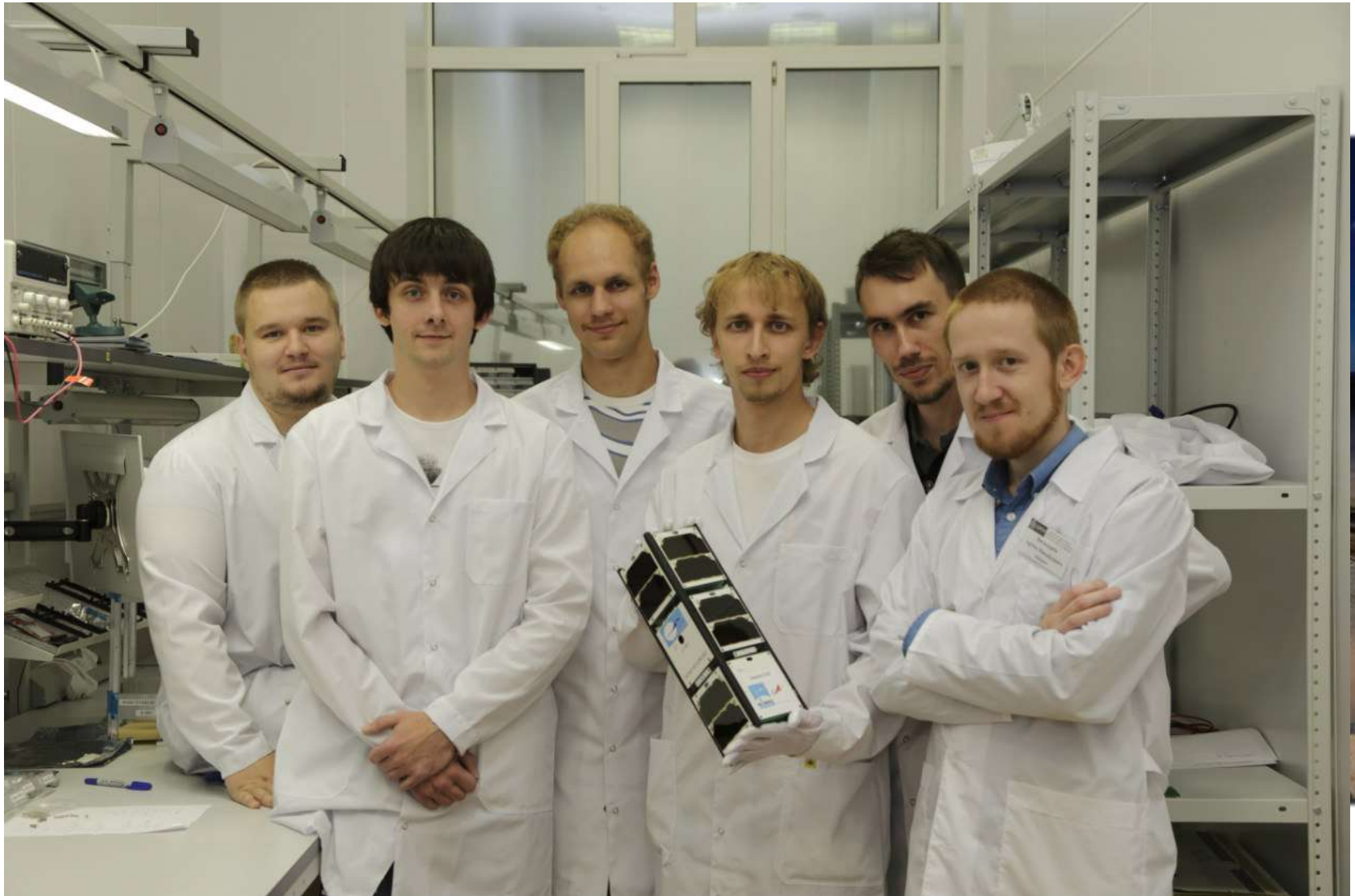


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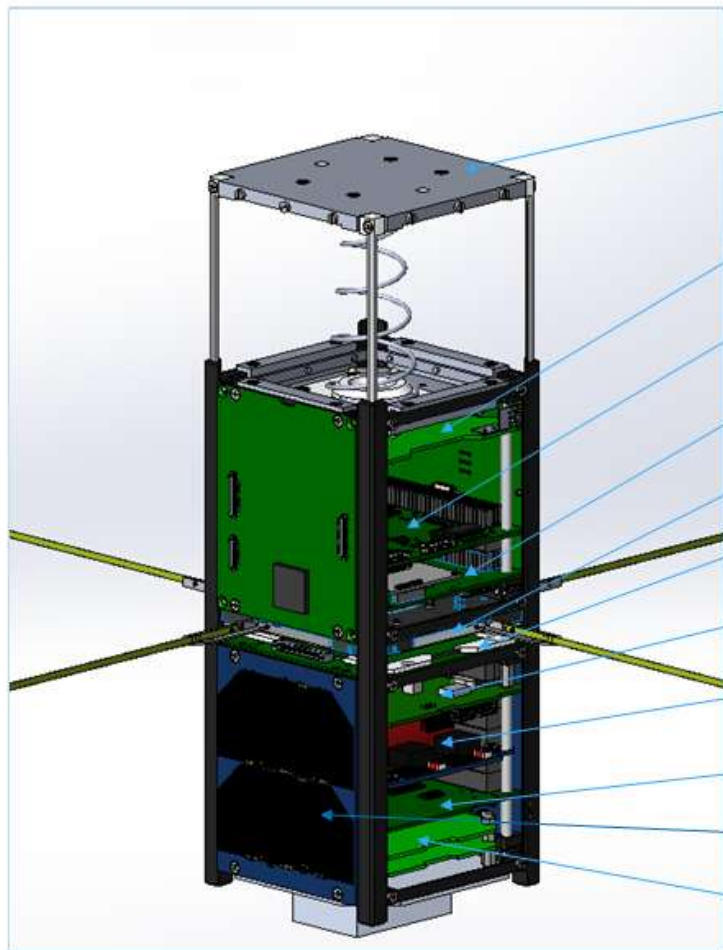


SAMSAT-218D

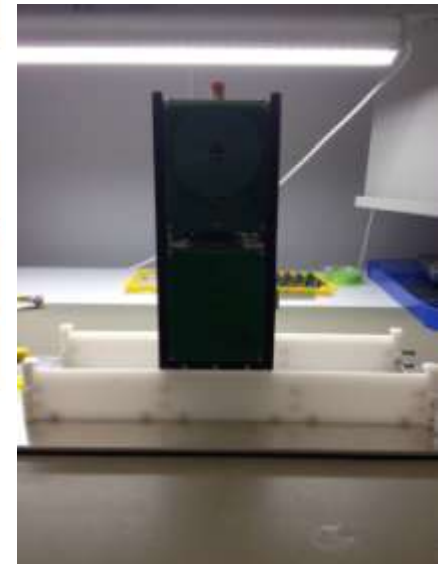




SAMSAT-QB50



- Aerodynamic Stabilizer
- Aerodynamic Stabilizer Opening System
- OBC
- GPS Receiver
- ANTS
- IGIS
- TRXUV
- EPS
- Payload Integration System
- Solar Panels
- FIPEX Payload



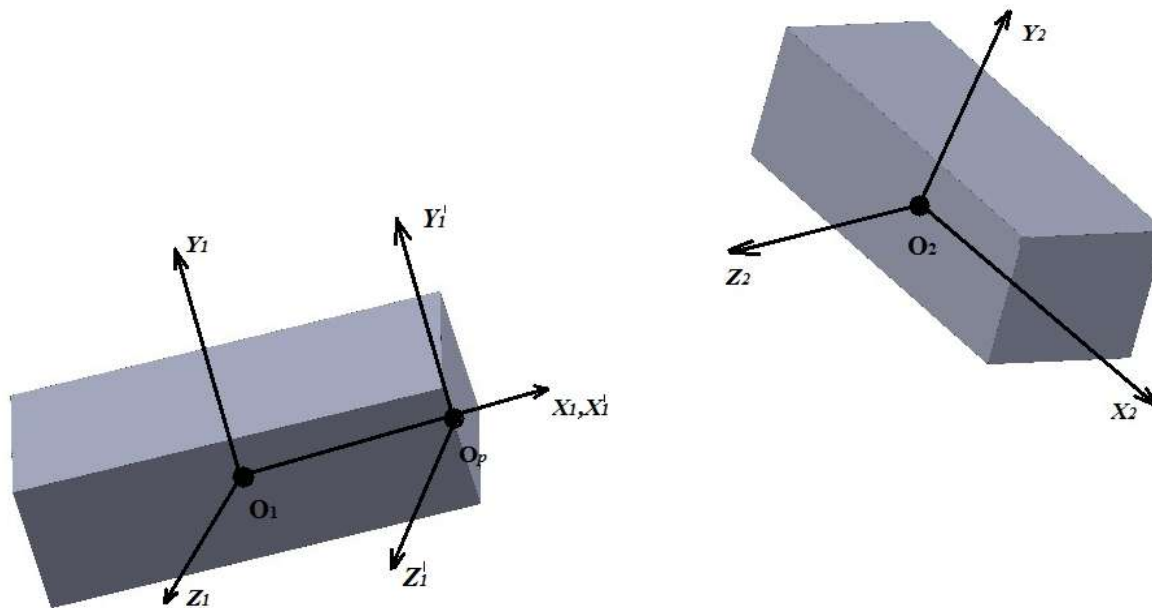


PROBLEM FORMULATION

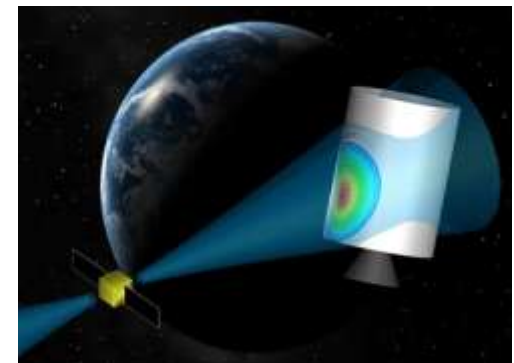
Currently, actively developed the theme of the formation flight.

In this flight are often used nanosatellite and one of the many methods of relative navigation into this groups may be navigation with video camera.

In this regard, the development of an independent algorithm for determining the navigation parameters from the onboard video equipment is an urgent task



Pic. 1. Used coordinate systems

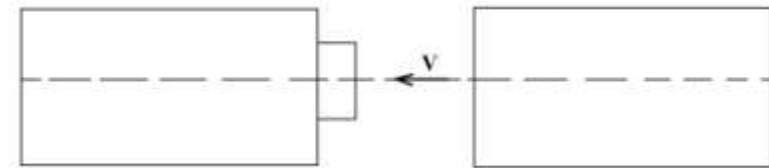




INITIAL CONDITIONS FOR MODELING

Table 1 – Parameters of the orbit and separation, mass, inertia and dimensional characteristics

Parameters of the orbit	
Height, км	300
Separation parameters	
Angular velocity, deg/s	
ω_x	0
ω_y	4
ω_z	4
Linear velocity, m/s	
V_x	0.01
V_y	0
V_z	0
Mass, inertia and dimensional characteristics	
Moments of inertia, kg/m ³	
I_x	0.0033
I_y, I_z	0.0103
Length, m	0.24
Mass, kg	2



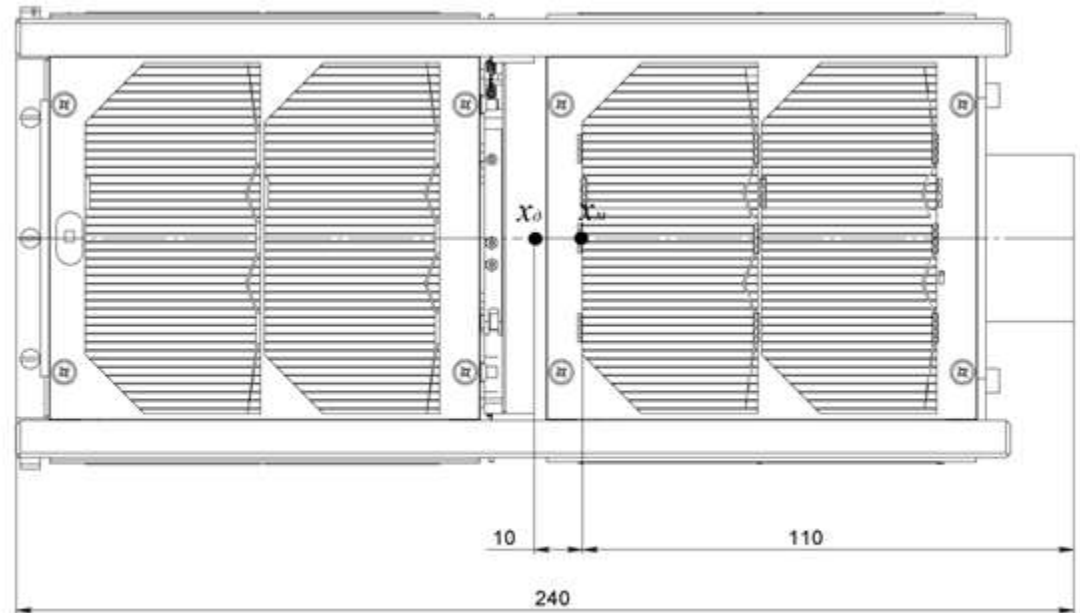
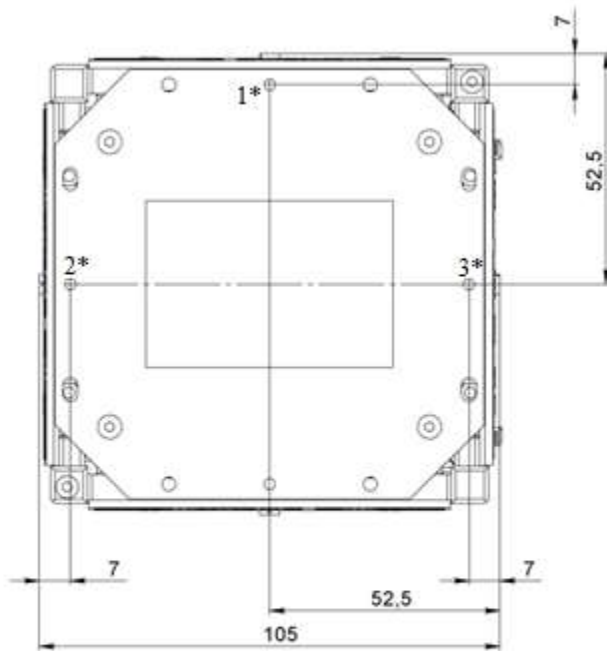
Pic. 2. Separation scheme

Table 2 – Position of the reference points(mm)

№	1	2	3
x_2	92	92	92
y_2	0	-45.5	45.5
z_2	45.5	0	0



MODELING OBJECT SKETCH

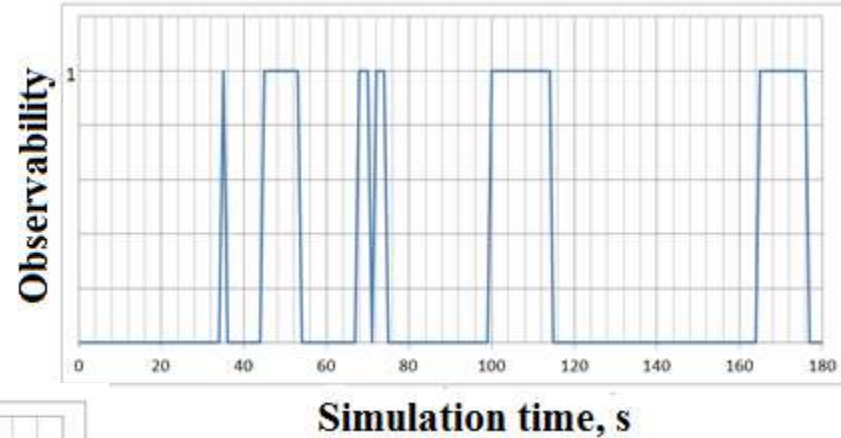


**-Reference points*



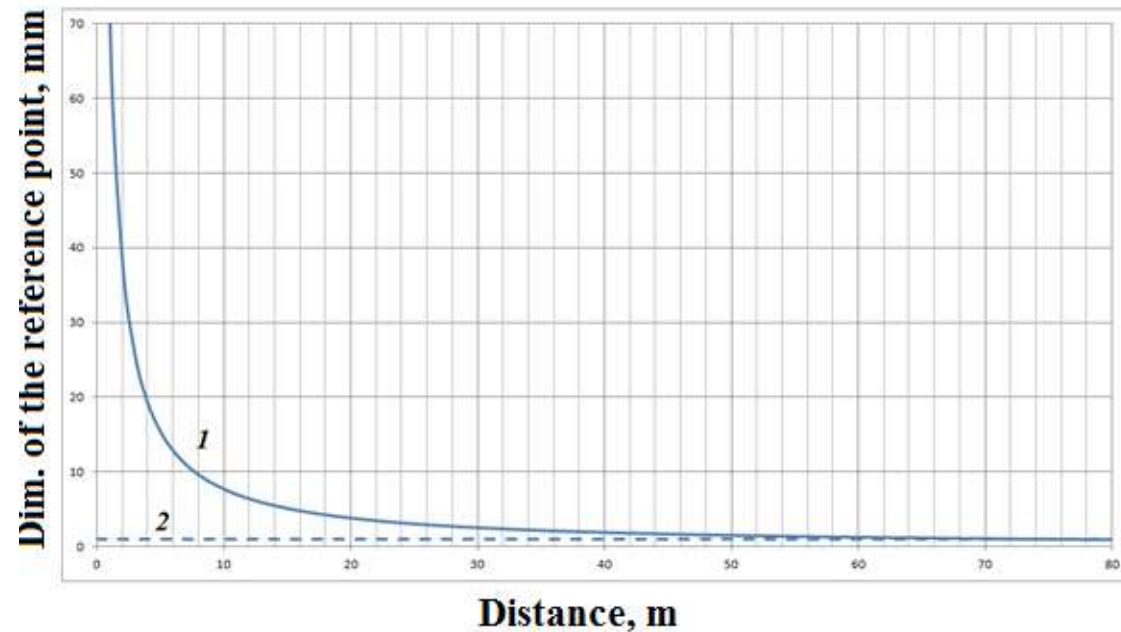
TECHNICAL ANALYSIS

1. The sensor has a rectangular shape with dimensions 4,8x3,6 mm;
2. The focal distance of the camera $f=29$ mm;
3. Camera resolution 640x480;
4. Linear size of the reference point $W=H=D=20$ mm;
5. The camera should make >2 snapshots.

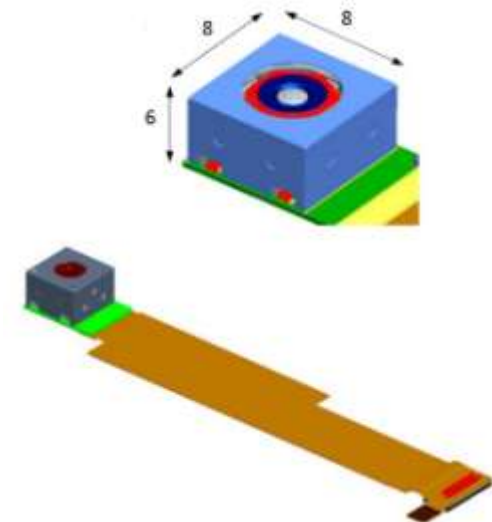


Simulation time, s

Pic. 3. Dependence observability of the time for the selected camera



Pic. 4. Dependence benchmark size of the distance between nanosatellites:
1 – dependence; 2 – conventional border

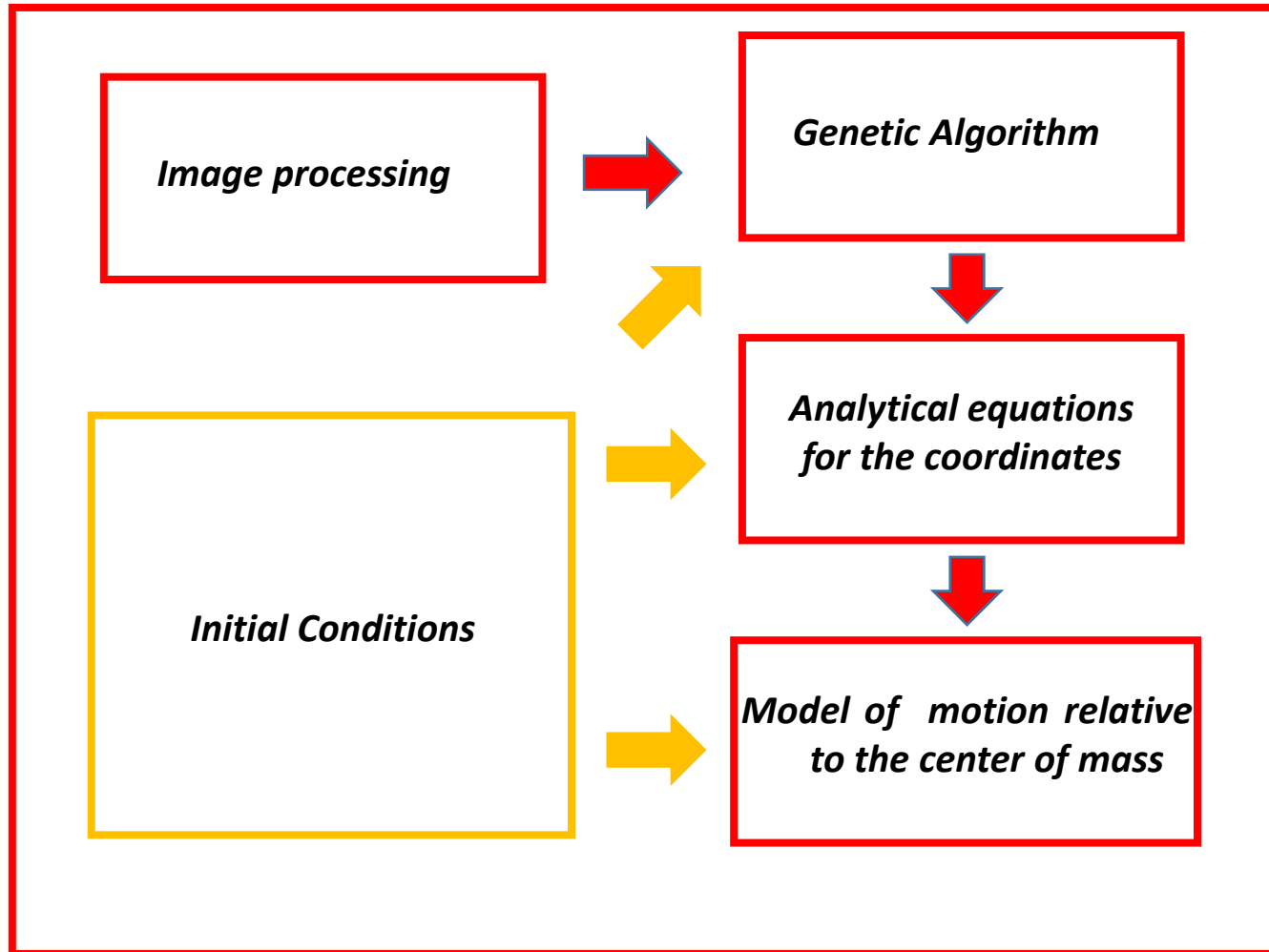


Pic. 5. Selected camera



ALGORITHM OF RELATIVE NAVIGATION

Algorithm structure:



Research method:

Method Monte Carlo



SIMULATION

1. Simulation time $t=60$ s;
2. Second nanosatellite observe by the first nanosatellite through all simulation time;
3. Initial population size $N=10000$;
4. Number of model experiments $n=400$;

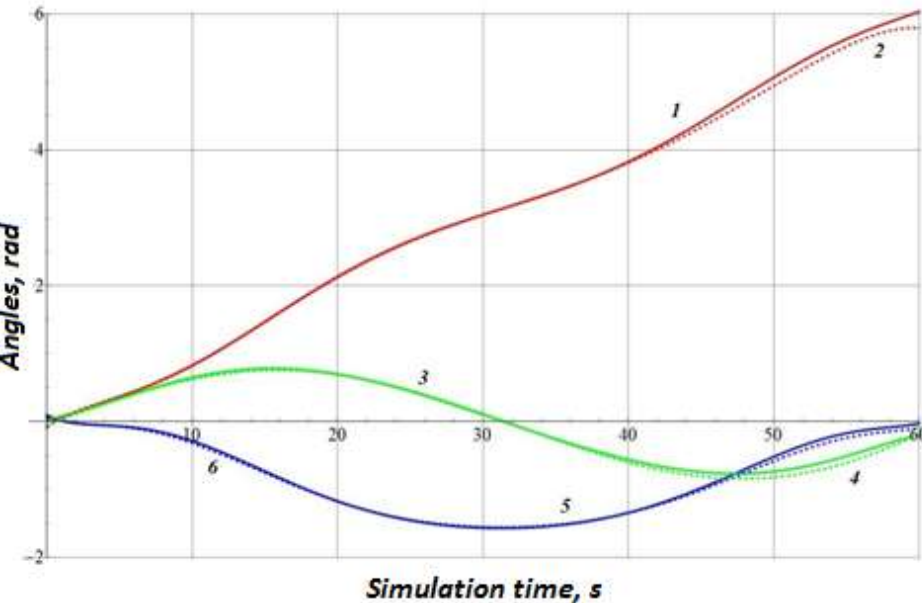


Рис 6. Изменение углов ориентации:

- 1 – initial angle ψ ; 2 – math. expectation of the angle ψ ;
3 – initial angle Θ ; 4 – math. expectation of the angle Θ ;
5 – initial angle φ ; 6 – math. expectation of the angle φ

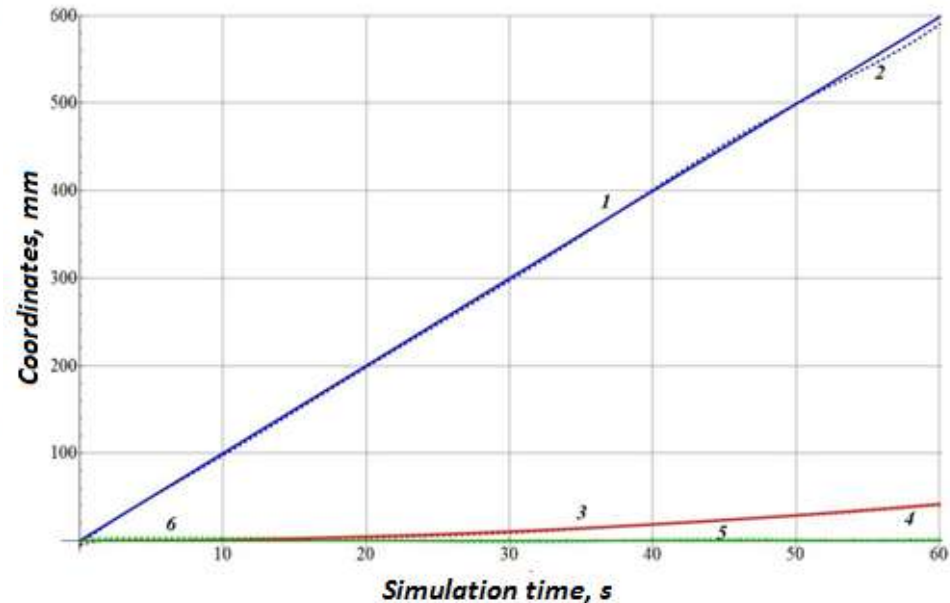


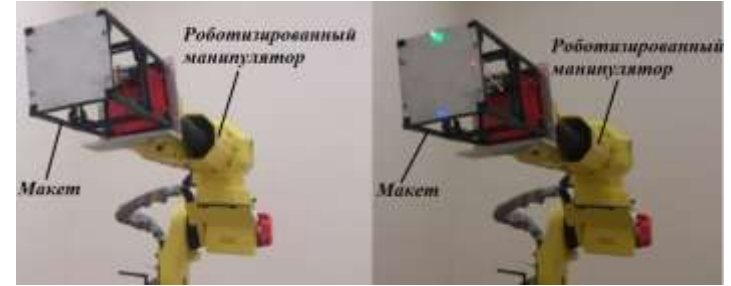
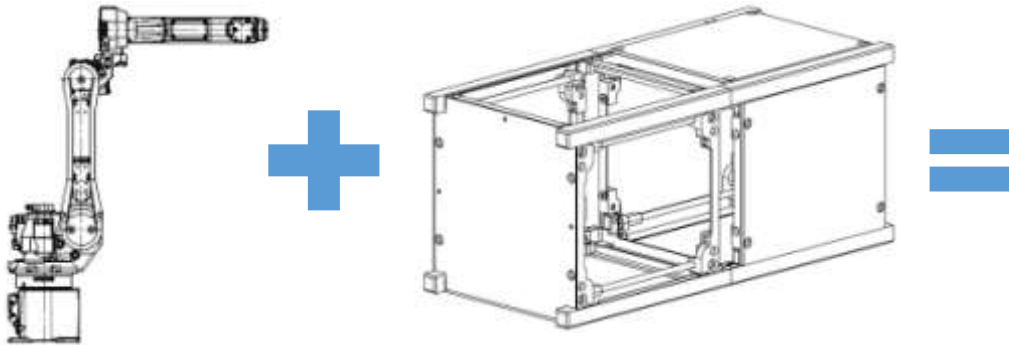
Рис. 7. Изменение координат центра масс:

- 1 – initial coordinate X_c ;
2 – math. expectation of the coordinate X_c ;
3 – initial coordinate Y_c ;
4 – math. expectation of the coordinate Y_c ;
5 – initial coordinate Z_c ;
6 – math. expectation of the coordinate Z_c

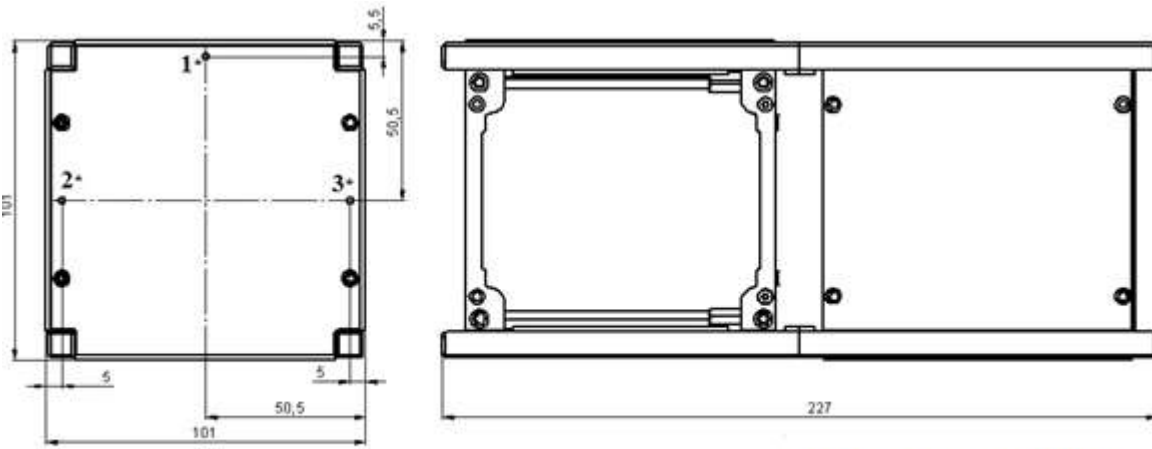




EXPERIMENTS



Pic. 8. Experimental installation



**-Reference points*

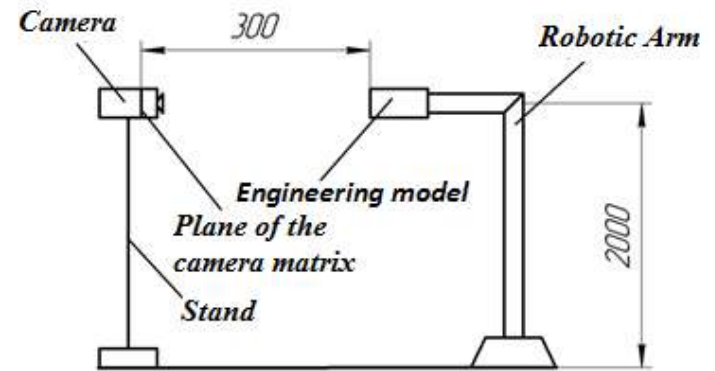


Pic. 9. Dimensions of the engineering model

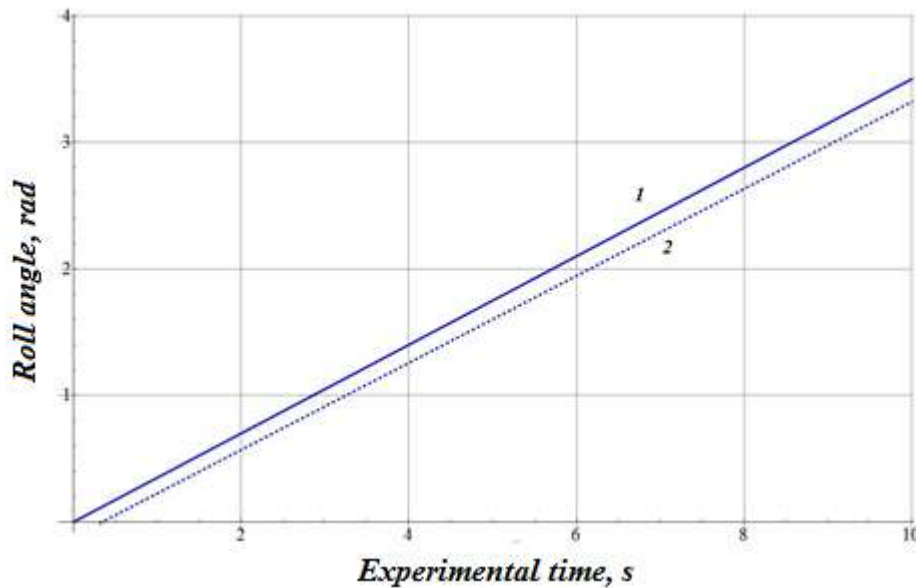


EXPERIMENTAL RESULTS

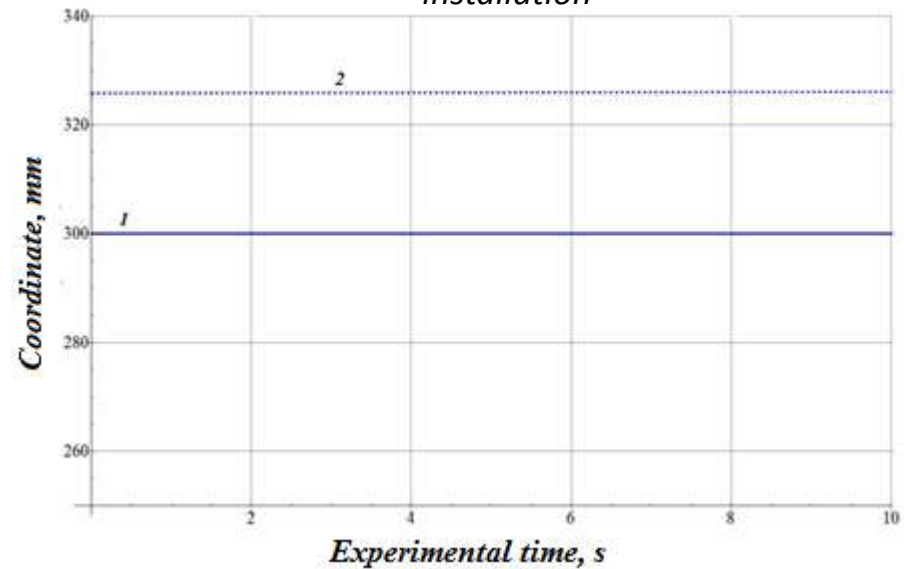
1. Modeling time $t=10$ s;
2. The distance between the camera and the subject is 300 mm;
3. Angular velocity $\omega_x = 22$ °/s, $\omega_y = 0$ °/s, $\omega_z = 0$ °/s;
4. Number of tests $n=10$.



Pic. 10. Scheme of the experimental installation



Pic. 11. Changing the roll angle:
1 – initial angle; 2 – math. expectation of the angle



Pic. 12. Changing the coordinate center of the mass:
1 – initial coord.; 2 – math. expectation of the coord.



- 1. An algorithm for finding relative orientation and position was developed;***
- 2. Spend a technical analysis of the selected optical camera;***
- 3. Was assessed performance of the algorithm on the modeling and on the experiment.***

А ТЫ СДЕЛАЛ



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34, Moskovskoye shosse, Samara, 443086, Russia
Tel.: +7 (846) 335-18-26, fax: +7 (846) 335-18-36
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